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REVISION DESCRIPTION:

Page: 41-08 REV 1: In Step 4 "a minimum of 1/16 [1.6 mm] gap" was "an approximately 1/16 [1.6 mm] gap."

In Step 4, deleted "Substitute washers as necessary..."

Added Step 5.

In Figure 3, "MIN. 1/16 [1.6 mm]" was "1/8 [3.175 mm]"



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It is the builder's choice as to whether to complete all steps for the left side before repeating those steps for the right side or to complete each step for both left and right sides before moving to the next step.

Step 1: Remove the Rod End Bearing from the Flap Assembly as shown in Figure 1.

<u>Step 2:</u> If necessary, install the flap as shown on Page 21-12, Figure 3.

Step 3: Trim the VA-256 Flap Pushrod as shown in Figure 2.

Step 4: Assemble the Flap Pushrod Assembly as shown in Figure 4. Use the Rod End Bearing removed in Step 1.

<u>Step 5:</u> Install the Flap Pushrod Assembly into the Flap Assembly as shown in Figure 3.











2X CM-4M 2X AN316-4R

FLAP ASSEMBLY



VA-256

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FIGURE 4: FLAP PUSHROD ASSEMBLY

CAUTION: Verify that the plugs are installed as shown on Page 18-09, Figure 2. It will not be possible to install the plugs after the wings are installed!

<u>Step 1:</u> Fabricate two drift pins by tapering the ends of two 3/8 [9.5 mm] diameter common hardware store bolts. The bolts must be at least 7 [177.8 mm] long. Finished pins should have at least 3 [76.2 mm] of full diameter shank and no trace of threads remaining.

<u>Step 2:</u> Lightly lubricate the drift pins prior to installation by spraying them with LPS #1, 2, 3, or a light coat of ordinary motor oil to avoid damaging the spar and bulkhead holes.

<u>Step 3</u>: Lightly lubricate the mating surfaces of the Main Spar Assembly and Bulkhead Assemblies with Boelube to ease installation. See Figure 1. The forward mating surfaces of the Main Spar and Bulkhead Assemblies are not shown in Figure 1, but correspond to the aft mating surfaces. Lightly lubricate these with Boelube as well.

NOTE: It is recommended that the builder have an assistant for Steps 4-9. Empty promises of food or money work well here.

<u>Step 4:</u> Slide the left wing into position. Ensure that the Main Spar Assembly is positioned between the F-01403 and F-01404 Bulkhead Assemblies, and that the Rear Spar Assembly is positioned between the two F-01405B Bulkhead Bars.

<u>Step 5:</u> Pin the wing in place by **GENTLY** driving the lubricated drift pins into the holes with a soft plastic hammer. Have an assistant gently rock the wing up and down, as well as fore and aft to aid in alignment while installing the pins.

NOTE: Complete Steps 6 thru 10 during the final installation of the wing.

CAUTION: Do not lubricate the bolt threads, doing so can result in incorrect bolt torque values.

Step 6: Lightly lubricate the larger diameter bolts as described in Step 2.

Install the larger diameter bolts and the associated nuts and washers as shown in Figure 1.

Do not fully torque the nuts yet.

Step 7: Remove the drift pins installed in Step 5.

Step 8: Lightly lubricate the smaller diameter bolts called out in Figure 1 as described in Step 2.

Install the smaller diameter bolts and the associated nuts and washers as shown in Figure 1.

Do not fully torque the nuts yet.

<u>Step 9:</u> Lightly lubricate the bolts used to secure the Rear Spar Assembly to the F-01405B Bulkhead Bars.

Install the bolts and associated hardware as shown in Figure 1.

Step 10: Fully torque all of the wing attach bolts. See Section 5.20 for proper torque values.







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Step 1: Install the left CS-00005 Stick to Torque Tube Pushrod as shown in Figure 1. Note the washer installed between the aft face of the outboard bearing of the CS-00005 Stick To Torque Tube Pushrod and the CS-00009 Aileron Torque Tube. Leave the nuts loose for now.

Step 2: Remove the CS-00008-L Control Stick from the CS-00007-L Control Stick Base.

Step 3: Place a digital level flat against the top flange of the F-01438 Cover Rib and zero the level.

Step 4: Place the digital level on the top of the CS-00007 Control Stick Base and move it to the angle specified in Figure 1. Use tape to immobilize the pitch system. This establishes the "pitch neutral" position.

Step 5: Ensure that the CS-00007 Control Stick Base is also in the "roll neutral" position by verifying that the right most edge of the control stick base is parallel to the face of the F-01438 Cover Rib. This establishes the "fully neutral" position.

Step 6: With the control stick base in the "fully neutral" position, verify that the ailerons are approximately 1/16 [1.6 mm] above the trail position with respect to the wingtip trailing edges.

NOTE: Newer revisions of the W-00026 Alignment Template will have a reflex ID hole as shown on Page 23-10, Figure 2. If the template does not have the reflex ID hole, use shims to rig the trailing edge of the Aileron to protrude approximately 1/16 [1.6 mm] to ABOVE the trailing edge of the template. The Ailerons must be in this slightly reflexed position to compensate for aerodynamic loads during flight.

Step 7: If the ailerons are not aligned in the correct trail position, adjust the aileron bellcranks per the instructions on Page 23-10, Step 2 through Step 5.

Step 8: Verify that the control stick base is in the "fully neutral" position, and adjust the rod-end bearings in the CS-00005 Stick To Torque Tube Pushrods until the ailerons are correctly aligned in the trail position.

Step 9: Reinstall the CS-00008-L Control Stick and remove the immobilization measures.

Step 10: Sweep the control stick through the full range of motion and check for interference.

Step 11: Tighten all hardware per Section 5.20.







Step 1: Ensure the flap motor is fully retracted.

<u>Step 2:</u> Connect the Flap Pushrod Assembly to the CS-00010 Flap Torque Arm as shown in Figure 1 Note the washers positioned between the Flap Pushrod Assembly and the CS-00010 Flap Torque Arm.

<u>Step 3:</u> Use a digital level to check the the Flap Assembly for correct alignment when fully retracted as shown in Figure 2. Adjust the length of the Flap Pushrod Assembly as necessary.

<u>Step 4:</u> Verify that there is an adequate gap between the fuselage and Flap Assembly throughout the full range of motion. There should be a minimum of 1/16 [1.6 mm] gap between the fuselage and FL-00002A Top Skin when the flap motor is fully retracted. See Figure 3.

Step 5: File the inboard edges of the Flap Assembly as required to obtain a sufficient gap as described in Step 4.



FIGURE 1: CONNECT FLAP PUSHROD ASSEMBLY







NOTE: Refer to Figure 1 for the following steps:

Step 1: Install the snap bushing into the F-14118B-L Lower Wing Root Fairing as shown in Figure 1.

Step 2: Install the F-14118B-L Lower Wing Root Fairing as shown on Page 41-05, Figure 1.

Step 3: Install the F-00039-L Fuel Vent as shown. Ensure the flare fits tightly against the fitting prior to tightening the nut. Never use the nut to draw the flared line into contact with the fitting.

Step 4: Mark the F-00039-L Fuel Vent approximately 1 [25.4 mm] below the bottom skin of the fuel tank.

Step 5: Remove the F-00039-L Fuel Vent and trim to the mark made in the previous step.

Step 6: Chamfer the F-00039-L Fuel Vent as shown.

Step 7: Reinstall the F-00039-L Fuel Vent as shown.





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