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REVISION DESCRIPTION:

Since all the KAI pages regarding W&B and the Installed Equipment list are similar, only a general description of the two types of changes made to various sections is provided below. These changes resulted from relocating the Weight and Balance Worksheet and the Installed Equipment List.

- a. The Installed Equipment List was moved to the Maintenance Manual from the POH.
- b. The Weight and Balance Worksheet and W&B-2 (the blank page that followed) were moved to the Maintenance Manual (Rev 9) from the Production Acceptance Procedure. The "WEIGHT AND BALANCE RECORD" page was page W&B-2 of the RV-12 Production Acceptance Procedures.

Example: Following is an example of how this change affected Section 44A Skyview Autopilot Servos.

"Step 6: In the RV-12 Maintenance Manual (MM) "INSTALLED EQUIPMENT LIST" table, mark the "DYNON AUTOPILOT SERVO" as installed in the "INSTALLED" column.

Enter 4.6 lb for "Weight", 101.5 in for "Location/Arm" and 467 in-lb "Moment" onto the same line as "DYNON AUTOPILOT SERVO".

NOTE: The remaining steps on this page are only applicable to a flying aircraft.

Step 7: In the RV-12 Pilot Operating Handbook (POH) "YOUR AIRPLANE" table, enter the new total values for the arm, weight, and moment of the installed equipment.

Step 8: In the RV-12 POH "YOUR AIRPLANE" table, recalculate and enter new values for the Empty Weight, Empty Moment, and Empty Arm.

Step 9: Make an entry, as calculated in the previous step, on the WEIGHT AND BALANCE RECORD page of the RV-12 Maintenance Manual as follows:

As of this date: _	//	/"
--------------------	----	----

was

"Step 6: On Page 4-2 SkyView and 4-4 of the RV-12 Pilot Operating Handbook: Enter the text "AUTOPILOT" onto a blank line under the "ITEM" column in both tables.

Enter 4.6 lb for "Weight", 101.5 in for "Location/Arm" and 467 in-lb "Moment" onto the same line as "AUTOPILOT" in both tables.

Recalculate and enter new values for the Empty Weight, Empty Moment and Empty Arm on Page 4-4 of the POH.

Step 7: Make an entry on page W&B-2 of the RV-12 Production Acceptance Procedures as follows:

As	of	this	date:	/ /	/	•
, 10	O.	ti iiO	auto.	 		



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The changes described above were applied to the following pages. The updated Rev level is listed:

40-15 (Rev 2)

43-11 (Rev 2)

43B-08 (Rev 2)

43C-07 (Rev 1)

44A-05 (Rev 4)

44B-10 (Rev 1)

53-12 (Rev 3)

53B-06 (Rev 1)

58-06 (Rev 1)

61-08 (Rev 1)

Additional changes were also made and are described below in the usual manner.

Page 44A-04 REV 3: Add "(WITH FLAPS UP AND WITH FLAPS DOWN)" to the WARNING.

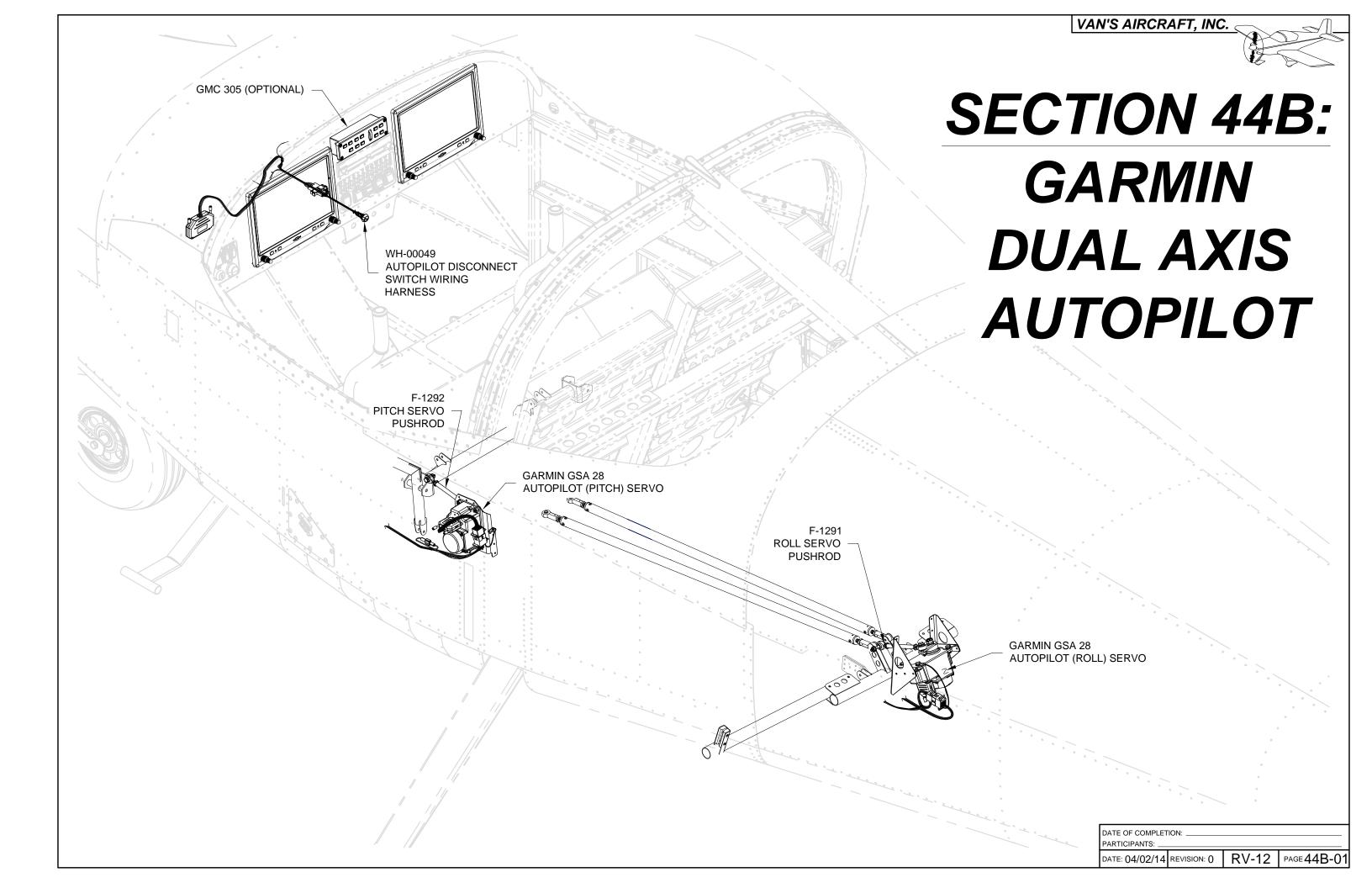
Page: 44B-03 REV 1: Add: "Step 4: Insert the bolt that will attach the Pitch Servo Pushrod Assembly to the arm of the Garmin GSA 28 Autopilot Servo. See Figure 2." Repaginate remaining steps.

Page 44B-06 REV 1: Add "(WITH FLAPS UP AND WITH FLAPS DOWN)" to the WARNING.

Page 44B-07 REV 2: Show additional cut lines for GMC 307 in Figure 2. Add "**(GMC 305)**" after hardware callouts in Figure 2.

Page: 53-10 REV 2: Deleted fuse amperage values shown in Figure 2, except for GPS ADSB, "2" AMP.

Page: 53B-05 REV 1: Deleted AMP values from fuses in Figure 2, except for GPS ADSB, "2" AMP.



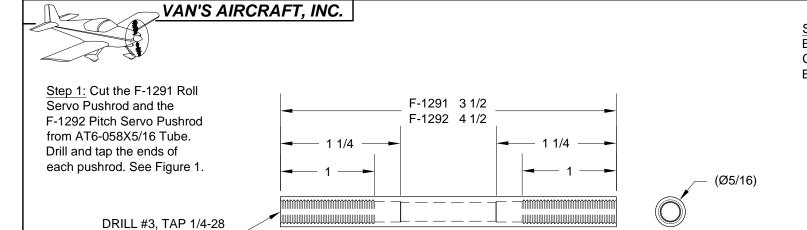


FIGURE 1: MAKING THE PUSHROD TUBES

Step 2: Assemble the F-1291 Roll Servo Pushrod and hardware as shown in Figure 2 to make the Roll Servo Pushrod Assembly. AN316-4R, 2 PLACES

BOTH ENDS

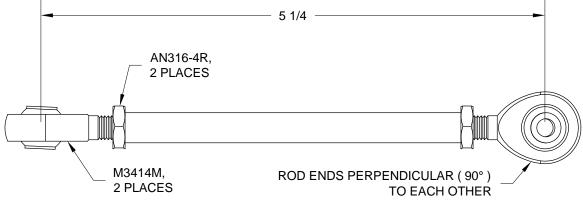


FIGURE 2: ROLL SERVO PUSHROD ASSEMBLY

Step 3: Assemble the F-1292 Pitch Servo Pushrod and hardware as shown in Figure 3 to make the Pitch Servo Pushrod Assembly.

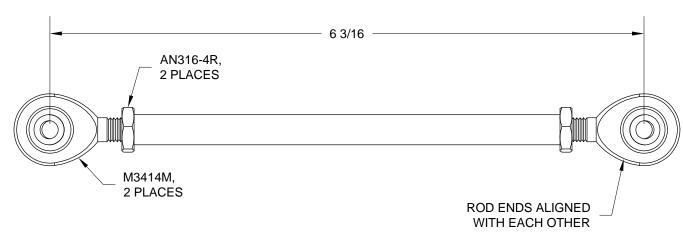


FIGURE 3: PITCH SERVO PUSHROD ASSEMBLY

Step 4: Remove the F-00081 ADAHRS Baggage Floor, the F-00083 Baggage Cover, and the F-1207F Baggage Bulkhead Corrugation. See Figure 4.

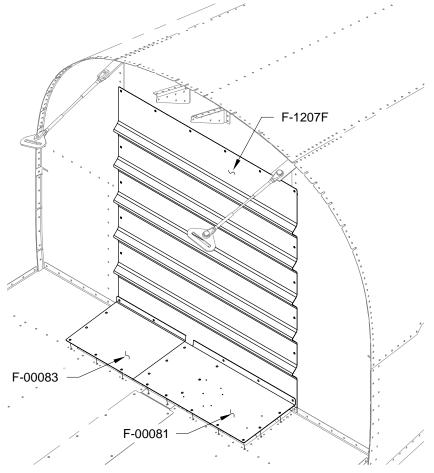


FIGURE 4: REMOVING ACCESS COVERS

Step 5: Remove the F-1227 Seat Ramp Cover. See Figure 5.

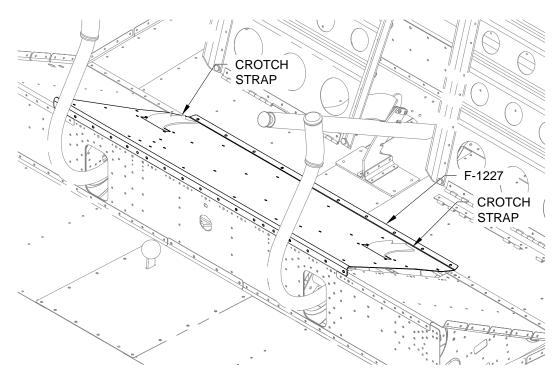


FIGURE 5: REMOVING THE SEAT RAMP COVER

NOTE: Some kits may include a pre-trimmed F-1269 Servo Doubler. If so, Step 1 through Step 3 does not apply. Set the F-00072 Trim Guide aside and proceed to Step 4.

Step 1: Attach the F-00072 Trim Guide to the previously installed F-1269 Servo Doubler using the hardware called out in Figure 1.

Step 2: Remove the excess material from the servo doubler and the F-1215-R Seat Rib to match the radius of the trim guide using a small die grinder.

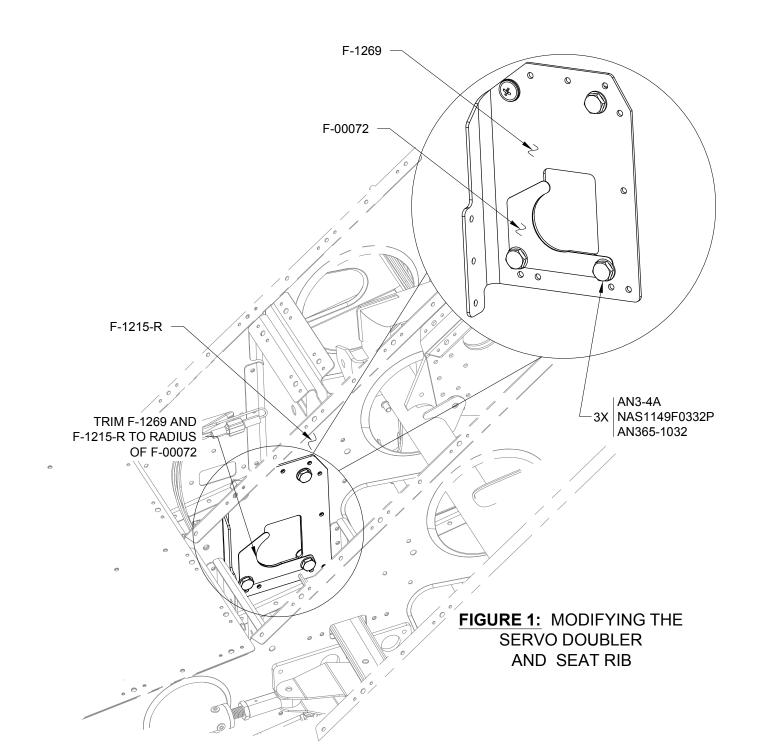
Step 3: Remove and discard the trim guide. Deburr any sharp edges resultant from grinding.

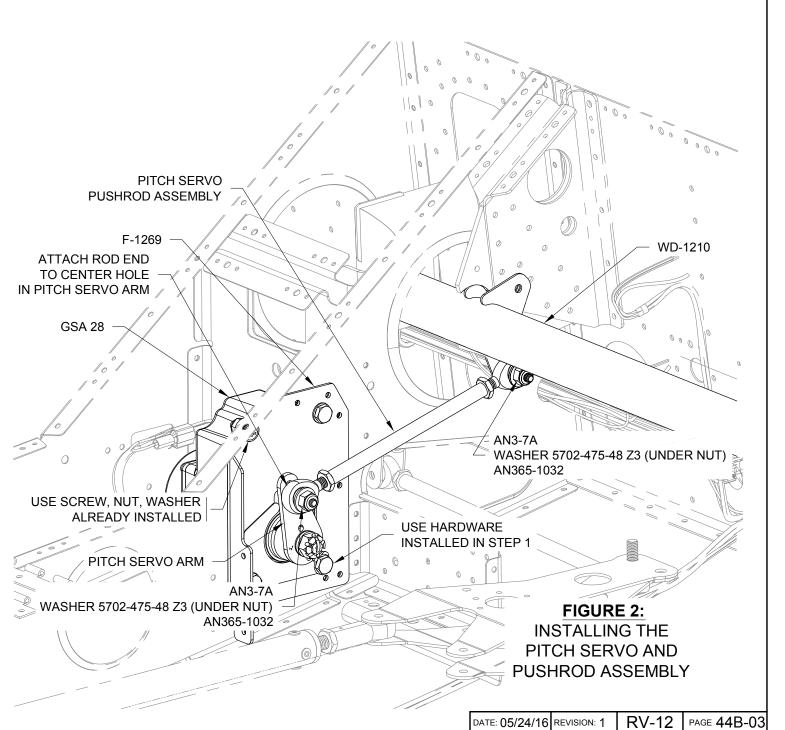
Step 4: Insert the bolt that will attach the Pitch Servo Pushrod Assembly to the pitch servo arm of the Garmin GSA 28 Autopilot Servo. See Figure 2.

<u>Step 5:</u> Install a Garmin GSA 28 Autopilot Servo to the F-1269 Servo Doubler. See Figure 2.

Step 6: Attach the Pitch Servo Pushrod Assembly to the Pitch Servo Arm as shown in Figure 2. Make certain the Pushrod Assembly is bolted to the center hole in the Pitch Servo Arm

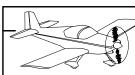
<u>Step 7:</u> Attach the other end of the Pitch Servo Pushrod Assembly to the tab on the lower surface of the WD-1210 Control Column. See Figure 2.





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NOTE: If your aircraft does not have a Molex connector on the autopilot servo wires coming from the WH-00045 Options Harness and WH-00046 Fuselage Harness, go to Page 44B-11, Step 4 for instructions on reconfiguring your harness to mate to the servo Molex connector.

Step 1: Mate the spade connector coming from the ES-00010 9-pin female Molex connector to the spade connector coming from the WH-00045 Options Harness. See Figure 1.

Use a tie-wrap to secure the mated spade connectors to the nearby wires from the WH-00046 Fuselage Harness and the WH-00045 Options Harness.

Step 2: Locate the WH-00082 Pitch Servo Harness with an 15-pin female D-Sub connector and an ES-00009 9-pin male Molex connector.

Attach the 15-pin female D-Sub connector labled "P-SERVO" to the 15-pin male D-Sub receptacle on the GSA 28 Autopilot Servo.

Connect the 9-pin male Molex connector to the 9 pin female Molex connector on the WH-00045 Options Harness.

Step 3: Secure the mated Molex connectors to the body of the pitch servo using two tie-wraps. See Figure 2.

Step 4: Locate the WH-F451 (BLK) wire coming from the ES-00010 9-pin Molex connector. Attach the ring terminal to the F-1215-R seat rib. See Figure 2.

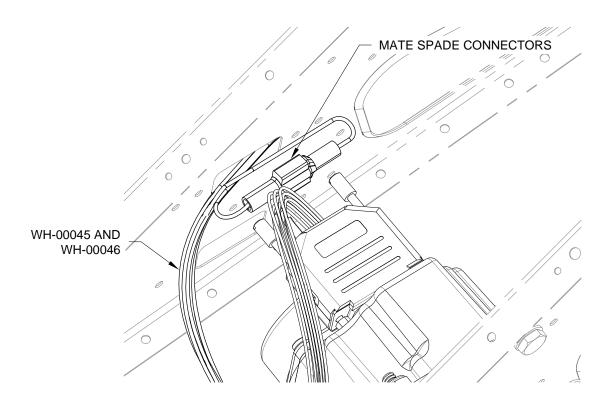


FIGURE 1: INSTALLING THE **SPADE CONNECTORS**

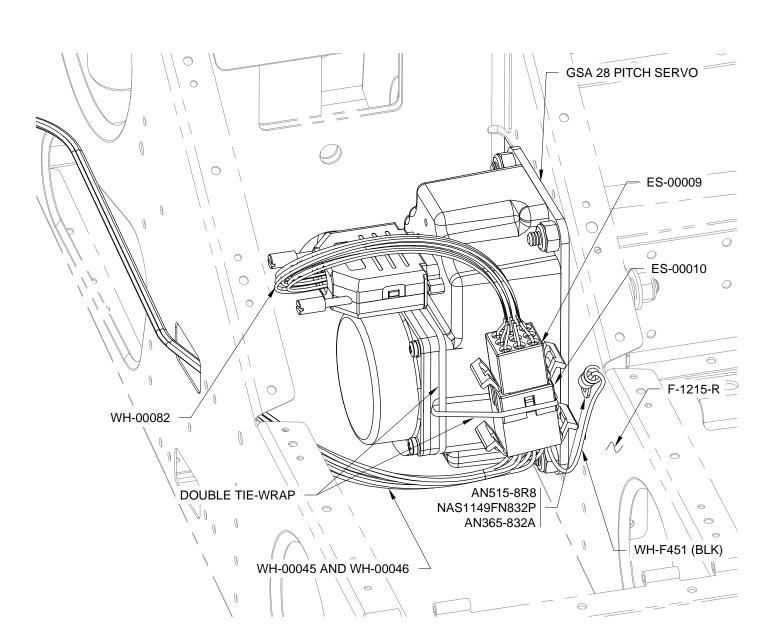


FIGURE 2: PITCH SERVO WIRING

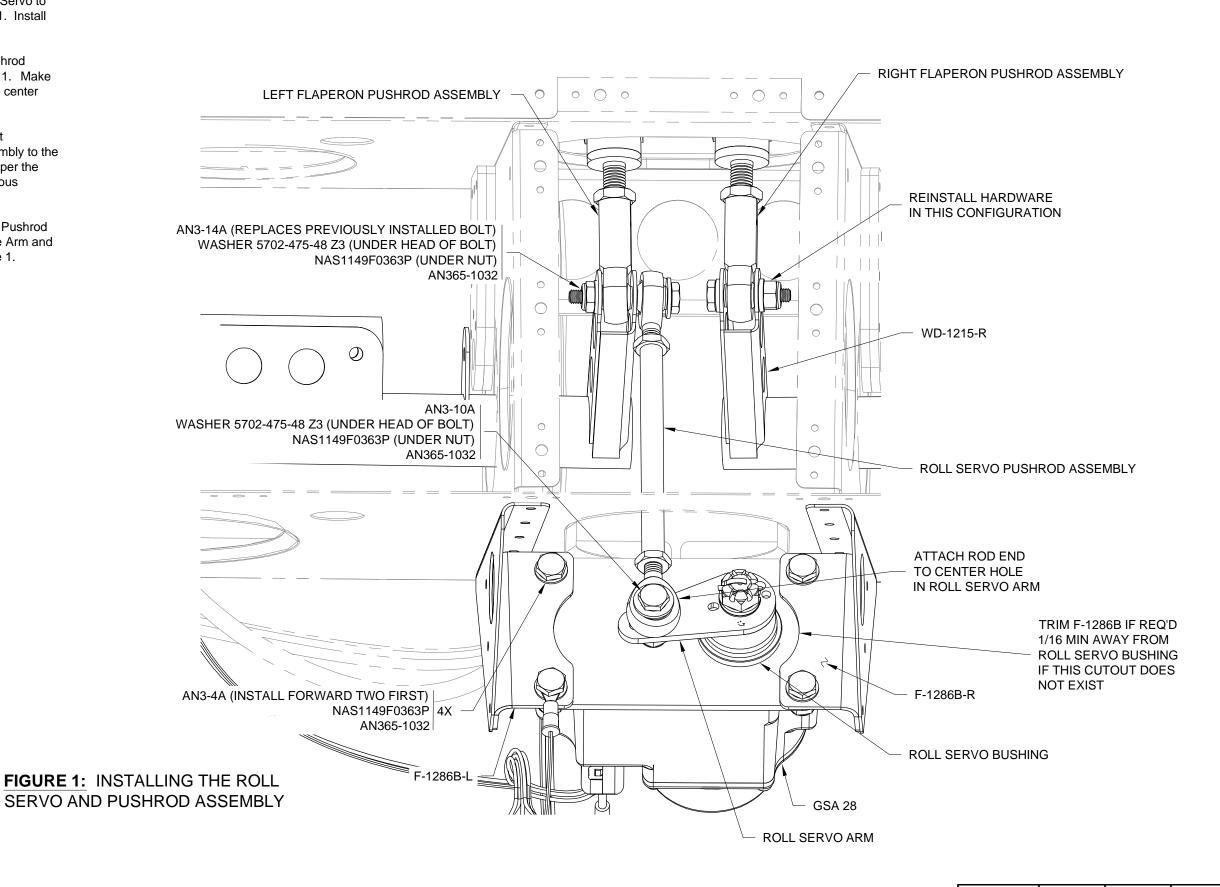
Step 1: Older F-1286B_Roll Servo Angles may not have a cutout for the bushing protruding from the top of the Garmin GSA 28 Autopilot Servo. Trim the angles if/as req'd. See Figure 1.

Step 2: Attach the Garmin GSA 28 Autopilot Servo to the F-1286B Roll Servo Angles. See Figure 1. Install the forward two bolts and hardware first.

Step 3: Locate and install the Roll Servo Pushrod Assembly to the Roll Servo Arm. See Figure 1. Make certain the Pushrod Assembly is bolted to the center hole in the Roll Servo Arm.

Step 4: Remove and reinstall the bolt and nut connecting the Right Flaperon Pushrod Assembly to the WD-1215-R Flaperon Torque Arm. Reinstall per the orientation shown in Figure 1. Maintain previous washer positions.

Step 5: Attach the free end of the Roll Servo Pushrod Assembly to the WD-1215-L Flaperon Torque Arm and Left Flaperon Pushrod Assembly. See Figure 1.



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NOTE: If your aircraft does not have a molex connector on the autopilot servo wires coming from the WH-00045 Options Harness, go to Page 44B-09, Step 1 for instructions on reconfiguring your harness to mate to the servo Molex connector.

TIE-WRAP

MOLEX

CONNECTORS

Step 1: Locate the WH-F450 (BLK) wire coming from the WH-00010 9-pin Molex connector attached to the WH-00045 Options Harness.

Secure the ring terminal under the head of the bolt shown in Figure 1.

Step 2: Locate the WH-00107 Roll Servo Harness.

Install the 15-pin D-Sub connector labled "R-SERVO" to the 15-pin D-sub receptacle on the Roll Servo.

Mate the ES-00009 9-pin Molex connector to the ES-00010 9-pin Molex connector on the WH-00045 Options

Step 3: Route the WH-F1163 (WHT) and WH-F1164 (BLK) wires (the long wires not terminated at the Molex connector) through the lightning hole in the F-1207B shown. Use the Aft Pitot Line as a routing guide. See Section 42N for instructions on installing these two wires.

Step 4: Use a tie-wrap to secure the mated Molex connectors to the lightning hole in the F-1286A-L Servo Bracket as shown in Figure 2. Make certain that the WH-F540 (BLK) wire will not interfere with the travel of the Roll Servo Arm.

WARNING: WHEN FINISHED INSTALLING THE GARMIN GSA 28 AUTOPILOT SERVOS, MOVE THE CONTROL STICK THROUGHOUT ITS ENTIRE RANGE OF TRAVEL (WITH FLAPS UP AND WITH FLAPS DOWN) MANY TIMES TO CHECK FOR AN OVER-CENTER CONDITION OF THE AUTOPILOT SERVOS (A CONDITION WHERE THE SERVO ARM AND PUSHROD BECOME CLOSE TO PARALLEL AND THE CONTROL SYSTEM LOCKS).

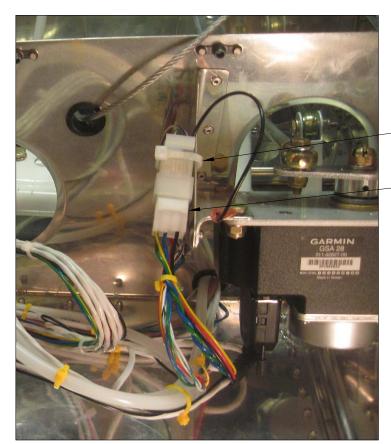
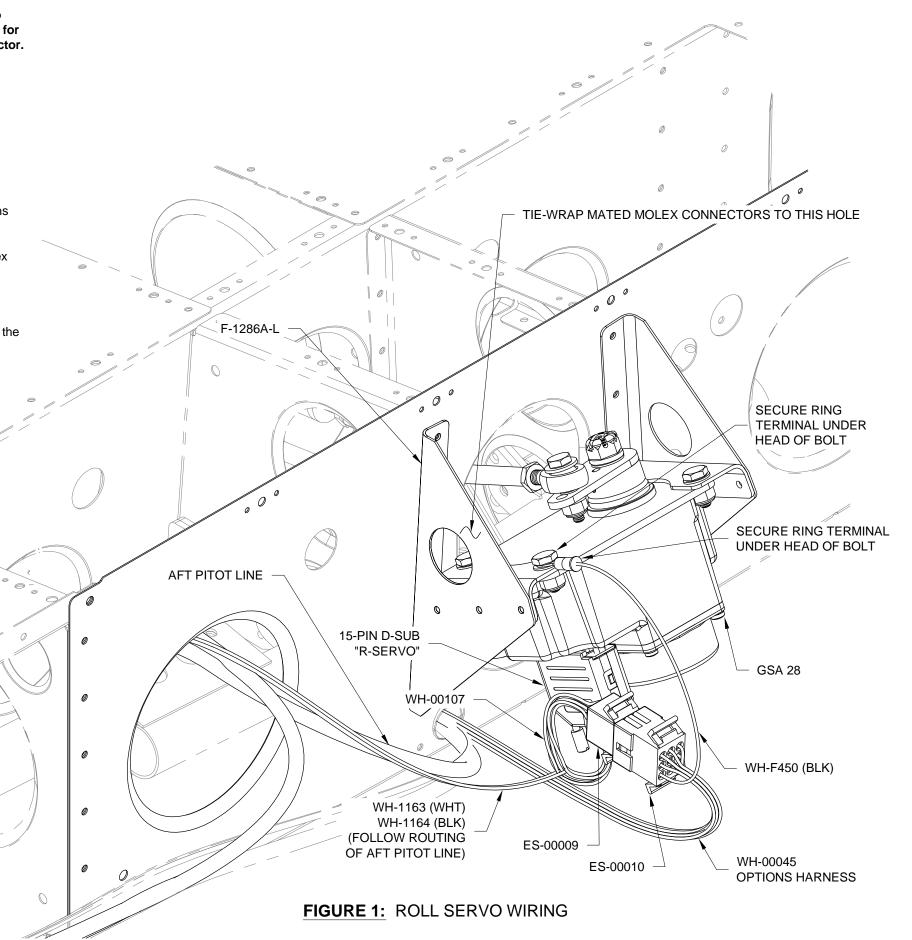


FIGURE 1: TIE-WRAPPING THE MOLEX CONNECTOR



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NOTE: The GMC 305 or GMC 307 is an optional installation. If not installing the GMC 305 or GMC 307 at this time proceed to page 44B-09.

NOTE: If installing the GMC 305 the F-00044G-1 Cntr Inst PnI GTR 200 GMC 305 is available with the opening already cut for order from Van's Aircraft upon request. If using this center panel, proceed to Step 7. This same panel may also be used with a GMC 307 and the modification listed in Step 5.

<u>Step 1:</u> Attach the F-00085 GMC 305 Trim Template to the F-00044-1 GTR 200 Center Panel with two of the panel attach screws. Confirm that the bottom edge is parallel to the opening for the GTR200. See Figure 1.

<u>Step 2:</u> Match-Drill the center panel using the trim template as a drill guide. See Figure 1. Place a cleco in each hole after it is drilled. The #27 and #40 holes only need to be drilled for a GMC 305. The GMC 307 does not use screws for attachment.

<u>Step 3:</u> Use a scribe or a fine tipped marker to mark trim lines on the center panel using the trim template as a guide. See Figure 1.

USE TWO #6 PANEL ATTACH SCREWS F-00085 F-00044-1 USE OUTSIDE EDGES OF SLOTS AS A GUIDE FOR MARKING THE TRIM LINES (+)0 0 MATCH DRILL #30 4PL 0 MATCH DRILL #27 4PL MATCH DRILL #40 8PL 0 0 0 0 0 \circ 0 0 \bigcirc

FIGURE 1: MARKING THE GTR 200 CENTER PANEL WITH THE GMC 305 TRIM TEMPLATE

<u>Step 4:</u> Remove the trim template from the center panel and replace the two panel attach screws.

Step 5: Use a straight edge to connect the trim lines marked on the center panel in Step 3. See Figure 2.

If installing a GMC 307 extend the upper edge of the cutout as shown in Figure 2.

<u>Step 6:</u> Remove the material between the trim lines on the center panel. See Figure 2. Use a rotary cutting tool to remove the material just up to the lines. Use a file to remove any left over material and to deburr the cut edges.

Step 7: Test fit the GMC 305 or GMC 307 to the new cut-out in the center panel.

Use a file to remove material from the cut-out in the center panel if/as necessary.

<u>Step 8 (GMC 305):</u> Machine countersink the eight #40 nutplate attach holes on the center panel to fit the head of an AN426AD3 rivet. See Figure 2.

Step 9 (GMC 305): Rivet four nutplates to the forward face of the center panel. See Figure 2.

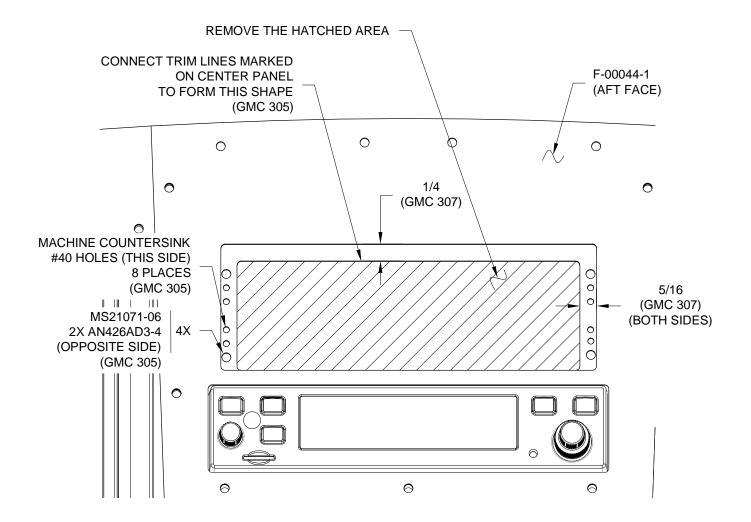
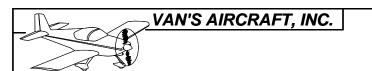


FIGURE 2: MAKING THE CUT-OUT

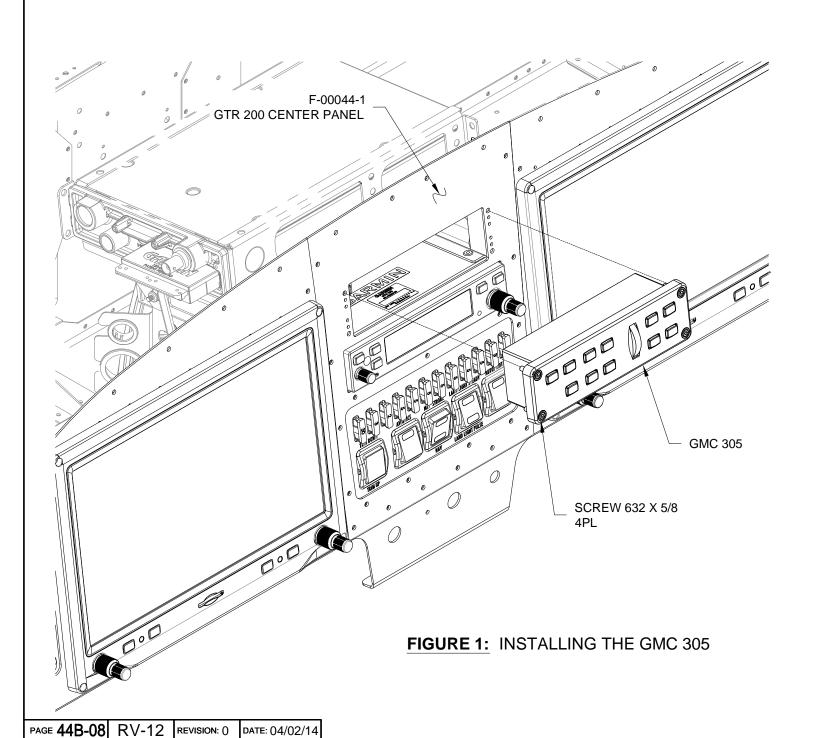


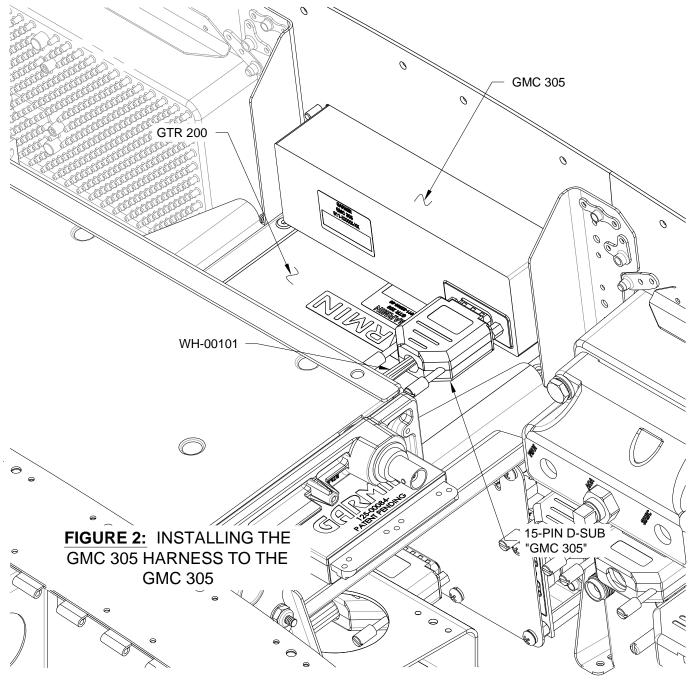
Step 1: Insert the GMC 305 into its cut-out in the F-00044-1 GTR 200 Center Panel. See Figure 1.

Step 2: Install the GMC 305 to the center panel using the attach hardware called out in Figure 1.

Step 3: Locate the previously installed WH-00101 G3X Harness and find the attached 15-pin high density D-Sub connector labled "GMC 305". Mate it to the 15-pin female D-Sub receptacle on the forward side of the GMC 305. See Figure 2.

Note the routing of the GMC 305 harness over the GTR 200.







Step 1: Re-install the F-1227 Seat Ramp Cover, F-00081 ADAHRS Baggage Floor, F-00083 Baggage Floor, and F-1207F Baggage Bulkhead Corrugation.

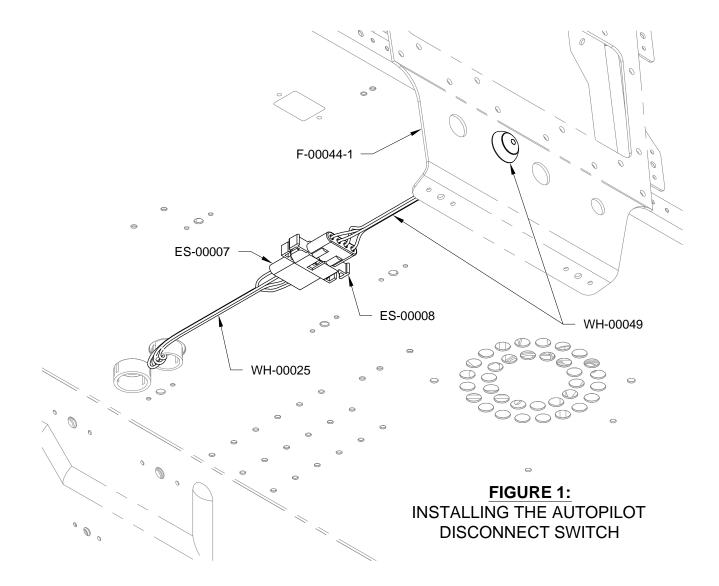
Step 2: Use a step-drill to enlarge the existing pilot hole in the F-00044-1 RV-12 GTR 200 Center Panel to 1/2 inch diameter. See Figure 1.

Deburr the hole.

Step 3: Install the WH-00049 Autopilot Disconnect Switch into the F-00044 RV-12 GTR 200 Center Panel as shown in Figure 1.

Insert the pins on the end of the autopilot disconnect switch wiring harness into a ES-00008 Molex Plug as shown in Figure 2.

Insert the ES-00007 Molex Receptacle into the Molex plug as shown in Figure 1.



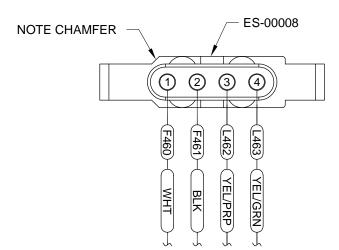


FIGURE 2: AUTOPILOT DISCONNECT DIAGRAM
(VIEW FROM WIRE INSERTION SIDE)

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Step 1: Navigate to the Van's Aircraft web site downloads page

Download and read the <u>READ ME.doc</u> document pertaining to performing a firmware update to the Garmin G3X Touch. Follow the instructions in this document to successfully install the .gca files.

Download and install the latest "RV-12 G3XT Base Configuration Rev X.gca" file.

Download and install from the same location the latest "RV-12 G3XT Autopilot Option.gca" file.

Finish the installation per the instructions in the Garmin installation guide. See the Garmin web site.

Step 2: Download the latest RV-12 overall Electrical Schematic from the Van's Aircraft web site.

NOTE: If installing the GMC 305 with the Autopilot Servos, skip Step 3 through Step 8 and proceed to Step 9.

NOTE: Any weight and balance information recorded for the aircraft must be updated. Depending on the state of your kit some steps may not be applicable.

Step 3: In the RV-12 Maintenance Manual (MM) "INSTALLED EQUIPMENT LIST" table, add "AUTOPILOT SERVOS" to the "ITEM" column. On the same line add a checkmark to the "INSTALLED" column.

Enter 3.1 lb for "Weight", 101.6 in for "Location/Arm" and 315 in-lb "Moment" onto the same line as "AUTOPILOT SERVOS".

In the RV-12 Pilot Operating Handbook (POH) "OPTIONAL EQUIPMENT LIST" table, add "AUTOPILOT SERVOS" to the "ITEM" column. On the same line add a checkmark to the "INSTALLED" column.

Enter 3.1 lb for "Weight", 101.6 in for "Location/Arm" and 315 in-lb "Moment" onto the same line as "AUTOPILOT SERVOS".

NOTE: Steps 4-6 on this page are only applicable if a final weight and balance as specified in the PAP has been completed.

Step 4: In the RV-12 Pilot Operating Handbook (POH) "YOUR AIRPLANE" table, enter the new total values for the arm, weight, and moment of the installed equipment.

Step 5: In the RV-12 POH "YOUR AIRPLANE" table, recalculate and enter new values for the Empty Weight, Empty Moment, and Empty Arm

<u>Step 6:</u> Make an entry, as calculated in the previous step, on the WEIGHT AND BALANCE RECORD page of the RV-12 Maintenance Manual as follows:

calculations resulting from the	ne installation of the Aut	opilot Servos Optional Kit.	
Revised Empty Weight: Revised Empty Moment:	lbs in-lbs		
Revised Empty Arm:	in	Signed:	

NOTE: Step 7 is only applicable for aircraft which have passed a final airworthiness inspection.

As of this date: ___/___ the following values represent current Weight and Balance

<u>Step 7 (ELSA):</u> Make an appropriate entry in the airframe logbook. See example below:

Installed the AUTOPILOT SERVOS option in accordance with Van's Aircraft KAI Section 44B and confirmed proper operation.

Signature	 Certificate #
Olgitaluic	 Ocitinoate #

Step 7 (SLSA): Complete the notification N 16-10-12 (available from the Van's Aircraft web site) corresponding to the AUTOPILOT SERVOS installation.

Step 8: Section complete.

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DATE: 05/24/16

NOTE: Any weight and balance information recorded for the aircraft must be updated. Depending on the state of your kit some steps may not be applicable.

Step 9: In the RV-12 Maintenance Manual (MM) "INSTALLED EQUIPMENT LIST" table, add "AUTOPILOT SERVOS AND GMC 305" to the "ITEM" column. On the same line add a checkmark to the "INSTALLED" column.

Enter 3.6 lb for "Weight", 95.3 in for "Location/Arm" and 343 in-lb "Moment" onto the same line as "AUTOPILOT SERVOS AND GMC 305".

NOTE: Steps 10-12 on this page are only applicable if a final weight and balance as specified in the PAP has been completed.

Step 10: In the RV-12 Pilot Operating Handbook (POH) "YOUR AIRPLANE" table, enter the new total values for the arm, weight, and moment of the installed equipment.

Step 11: In the RV-12 POH "YOUR AIRPLANE" table, recalculate and enter new values for the Empty Weight, Empty Moment, and Empty Arm.

Step 12: Make an entry, as calculated in the previous step, on the WEIGHT AND BALANCE RECORD page of the RV-12 Maintenance Manual as follows:

	O .	present current Weight and Balance	
calculations resulting from the	installation of the Aut	opilot Servos and GMC 305 Optional	Kit.
Revised Empty Weight:	lbs		
Revised Empty Moment:	in-lbs		
Revised Empty Arm:	in	Signed:	
. ,		3	
NOTE: The remaining steps on thi inspection.	s page are only app	licable for aircraft which have passe	ed a final airworthiness

Step 13 (ELSA): Make an appropriate entry in the airframe logbook. See example below:

Installed the AUTOPILOT SERVOS AND GMC 305 option in accordance with Van's Aircraft KAI Section 44B and confirmed proper operation.

Signature _____ Certificate # ____

Step 13 (SLSA): Complete the notification N 16-10-12 (available from the Van's Aircraft web site) corresponding to the AUTOPILOT SERVOS AND GMC 305 installation.

Step 14: Section complete.

NOTE: This page applies to kits with a date of purchase of the empennage, wing, fuselage or finish kit prior to September 2011.

NOTE: Step 1 through Step 3 applies to wire ends near the Roll Servo. Step 4 through Step 9 applies to wire ends near the Pitch Servo.

<u>Step 1</u>: Cut the spade connectors off the ends of the seven WH-B170 Autopilot Harness wires. (These spade connectors were installed per Page 31-09, Step 11.)

NOTE: See chapter 5W Open Barrel Crimp.

Step 2: Strip the ends of the seven WH-B170 Autopilot Harness wires and crimp an ES-00005 Molex Pin, .093" (18-22) onto the end of each wire.

<u>Step 3:</u> Insert the seven WH-B170 Autopilot Harness wires into an ES-00010 Molex Plug, 9 Position as shown in Figure 1. Numbers identifying the wire positions are molded into the back of the connector.

Continue the installation process by returning to Page 44B-06, Step 1.

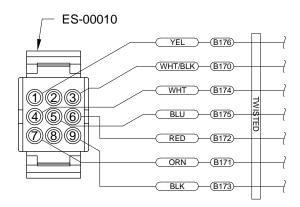


FIGURE 1: WH-B170 CONNECTION TO ROLL SERVO (VIEW FROM WIRE INSERTION SIDE)

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NOTE: <u>DO NOT</u> cut the male spade connector off the red WH-RV12-OPTIONAL harnessand WH-B170 Autopilot Harness wires.

Do not cut the female spade connectors off the ends of the BLU and YEL WH-RV12-OPTIONAL harness wires and WH-B170 Autopilot Harness wires.

<u>Step 4</u>: Cut the female spade connectors off the ends of the WHT and BLK WH-RV12-OPTIONAL harness wires and WH-B170 Autopilot Harness wires. (Each spade connector is crimped onto two wires. These spade connectors were installed per Page 31-09, Step 8.)

Cut the male spade connector off the ORN WH-RV12-OPTIONAL harness and WH-B170 Autopilot Harness wires.

Cut the male spade connector off the WHT/BLK WH-RV12-OPTIONAL harness and WH-B170 Autopilot Harness wires.

NOTE: See chapter 5W Open Barrel Crimp.

Step 5: Strip the ends of the twelve WH-RV12-OPTIONAL/WH-B170 Autopilot Harness wires. Twist together the stripped ends of each pair of like colored wire and crimp an ES-00003 Molex Pin, .093" (14-20) onto the end of each wire pair.

<u>Step 6:</u> Insert the six WH-RV12-OPTIONAL/WH-B170 Autopilot Harness wire pairs into an ES-00010 Molex Plug, 9 Position as shown in Figure 2. Numbers identifying the wire positions are molded into the back of the connector.

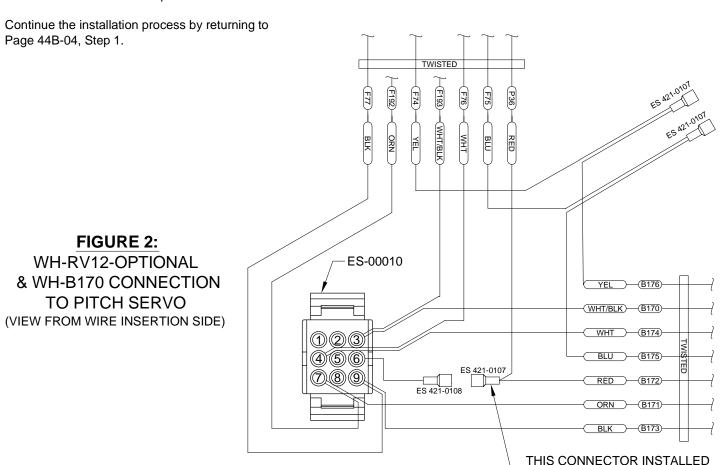
Step 7: Cut a piece of 22 gauge wire 3 1/2 inches long.

Strip one end and crimp on an ES-00005 Molex Pin, .093" (18-22).

Strip the other end and crimp on an ES 421-108 Female Spade Connector.

Insert the Molex pin end of the wire into position 6 of the ES-00010 Molex Plug, 9 Position as shown in Figure 2.

<u>Step 8:</u> Use a tie-wrap to secure the BLU and YEL wires (with spade connectors still attached) to the WH-RV12-OPTIONAL harness and WH-B170 Autopilot harness wires.



DATE: 04/02/14 REVISION: 0

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ON PAGE 31-09, STEP 8

